

### **GLAST Large Area Telescope:**

**LAT System Engineering** 

Pat Hascall SLAC System Engineering



### **Topics**

- Action Item Status
- Technical Baseline Management
- Issues
- Interface Control Documentation
- RFA Closure
- Key Metrics
- Risk Management



### **Monthly Action Item Status**

Action Item ID	Actionee	Description	Status
7-30-03-008	Jerry Clinton	Define and maintain the production readiness/execution plan to include vendor selection and associated schedule to ensure unit availability dates are met	OPEN: Draft production plan completed & provided to GSFC. Refinement required as vendors are selected. Update provided early December, 2003. Next update and process for update: TBD. Schedules for TEM/TPS provided to B.Graf, action to be closed when similar schedules are provided for the rest of the boxes. PDU schedule to be delivered today, GASU schedule by end of week



# Technical Baseline: Flight Drawing Release

- Status details (DAQ reported separately)
  - Tracker
    - 141 of 141 completed (total is 15 over original plan)
  - ACD
    - One assembly drawing remains, no impact to delivery
  - Mech
    - Completed 57 of 78 (total is 19 part over original plan)
    - Have added 15 MLI drawings
  - Design Integration
    - Major drawings: 1 of 5 signed off
    - MLI transferred to Mech



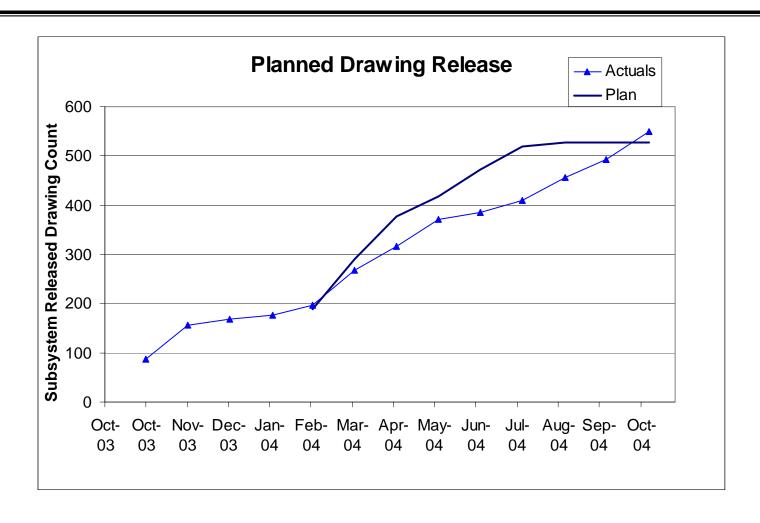
### **Technical Baseline: DAQ Flight Drawing Release**

• The table provides an update to the DAQ plan, and provides status on the progress of drawings that are not yet released

Group	Total	In Config Control	To Go	Percent Complete for To Go items	Need Date
TEM/TPS	48	48	0	100	9/20
PDU	35	31	4	100	10/20
GASU	63	41	22	35	11/3
EPU/SIU	63	35	36	69	11/3
Harness	28	21	7	14	12/3
Brackets	28	19	9	100	12/3
Heater Control Box	20	1	19	47	12/10
Total	285	196	97		



### Flight Drawing Release (as of 31 Oct)





### **Cumulative Released Drawing Metrics as of 31 Oct**

Subsystem	Jan-04	Feb-04	Mar-04	Apr-04	May-04	Jun-04	Jul-04	Aug-04	Sep-04	Oct-04
Tracker Plan	62	64	81	127	129	130	130	130	130	130
Actuals	62	64	81	82	117	129	139	139	139	141
ACD										
Plan	47	57	99	105	105	105	105	105	105	105
Actuals	47	57	99	99	103	103	103	104	104	104
Cal Plan	28	28	36	38	38	38	38	38	38	38
Actuals	28	28	36	38	41	41	41	41	41	41
DAQ										
Plan	0	0	30	50	78	125	172	181	181	181
Actuals	0	8	8	45	54	58	72	119	144	196
Mechanical Plan	39	39	43	52	54	59	59	59	59	59
Actuals	39	39	43	53	55	58	55	54	54	57
Assembly Plan	0	0	0	5	13	15	15	15	15	15
Actuals	0	0	0	0	0	0	0	0	10	10
Total Plan	176	188	289	377	417	472	519	528	528	528
Actuals	176	196	267	317	370	386	410	457	492	549



### **Issues**

No.	Description	Status	Due Date	Actionee
3	Technical baseline:	-All drawings to be under CM prior	Weekly Review	P. Hascall
	Flight Drawing	to flight build		
	release	-Flight drawing release plan		
		generated and statused weekly		
22	ASIC radiation	Radiation testing scheduled for	30 April ->June->	Sadrozinksi
	sensitivity testing	completion. 2 parts in work at		
	completion	SLAC, 2 in work at GSFC		
24	No plans to conduct	Looking at an EMI/EMC test to be	30 Sept	Blanchette
	Tracker Subsystem	performed after Tracker delivery		
	EMI/EMC	but before integration.		
25	EEPROM read/write	Gathered other program experience.	10/1 <b>-&gt;11/5</b>	Haller
	issues during LAT	Established mitigation options for	decision for	
	EM board testing	SIB and RAD750. Current plan	RAD750 PROM	
		retains baseline file management		
		system for SIB.		
27	PMT failures during	Root cause identified, alternate		Thompson
	thermal cycles	mounting method selected and		
		qualification underway		
		First 10 built, going into test		



# **Issues (continued)**

No.	Description	Status	Due Date	Actionee
28	Humiseal Conformal	GSFC QE review 8/31. <b>Implemented</b>	11/5 for	Rich
	Coat lifting	100% screening, rework process in	process defn.	
		release cycle,		
		Test for propagation in process		
29	Pitch adapter trace	Continuing to screen original design,	11/15	Rich
	cracks in bend	Have samples of masked pitch		
		adapter, waiting for samples of		
		selective plating		
30	Kapton to tungsten	Tiger team led by N. Johnson <b>GSFC</b>		N Johnson
	foil bonding	providing processed flight tiles		
31	Tracker flex cable	Process change implemented. First	10/15/04-	Rich
	coupon failures	coupon passed at GSFC. Coupons	>11/5	
	_	from flight panels in evaluation		
32	Tracker wire bond	Evaluating root cause. Potentially		R. Johnson
	breaks (heavy trays)	delete encapsulation. <b>Reviewing</b>		
		alternate methods		
33	FPGA failures	Tracking Aerospace Tiger Team	11/1/04	Haller
		efforts. Ordered alternate parts (UMC).		
35	Reliability	FMEAs done, reviews with	12/31/04	DiVenti
	assessments not	Subsystems started		
	completed			
36	SIIS capability to	Identifying shortfalls and plans to	Summer 05	Horwitz
	support I&T	mitigate with supplemental EGSE.		
		Needed for initial LAT		
		Comprehensive test		



### **Interface Management**



#### Interface Document Status

- SC-LAT ICD ICN Status
  - LAT signed this month
    - ICN-74 LAT Regulated OAP
    - ICN-75 LAT Daily Data Volume
  - Currently under signature review
    - ICN-76 Spare Discrete Mon for SIU Boot Status
  - Currently in draft or revision
    - None
- Internal LAT ICD's and IDD's
  - Signed this month
    - ACD-LAT ICD
  - Currently in update
    - Electronics-LAT ICD (Comments being incorporated as they are received)
    - TKR-LAT ICD's
    - CAL-LAT ICD



### **GFE Deliverables/Receivables**

#### LAT GFE Deliverables

Oct: None

Nov: ISIS (was October)

Dec: None Scheduled

Jan: None Scheduled

#### LAT GFE Receivables

Oct: None

Nov: SC-LAT Test Flexures (was October);

SIIS (was October)

Dec: None Scheduled

Jan: None Scheduled



### **RFA Closure**

- 37 CDR RFAs total, submitted 36 answers
  - Still working Radiator MGSE response
- Peer review RFAs
  - 177 pre CDR RFAs, one outstanding
    - ACD handling plan for blankets
  - 21 post CDR RFAs, 1 outstanding
    - X-LAT thermal test approach answer drafted and in review



# **Key Design Metrics**



#### **LAT Mass Status**

LAT Mass Status Report LAT-TD-00564-10

LAT Mass Status

Martin Nordby

LAT-TD-00564-10

Effective Date: 15-Sep-04

Print Date: 15-Sep-04

#### Sep-04

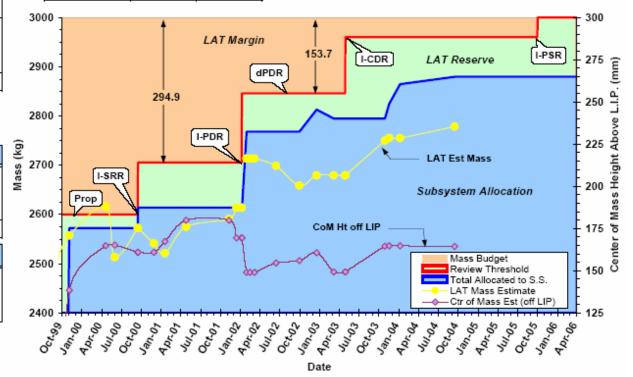
Mass (kg)	Estimate	Alloc.
TKR	514.0	510.0
CAL	1374.3	1440.0
ACD	286.2	295.0
Mech	366.6	386.6
Elec	230.4	240.0
Systems	7.0	8.0
LAT Total	2778.5	2879.6
Rsrv/Margin	221.5	
Rsrv/Margin*	8.0%	
Allocation		3000.0

<sup>\*</sup> AIAA G-020 recommended min reserve = 5.2% Allocations per latest mass CCB on 18 June 2004

Center of Mass (mm)					
CMx	-1.22	-20 < CMx < 20			
CMy	-0.89	-20 < CMy < 20			
CMz	-72.55	CMz < -51.2			
Ht off LIP	163.65	Ht < 185			

Second Moment of Inertia (kg-m²)					
lxx	1084.5	1500.0			
lyy	1032.1	1500.0			
Izz	1410.8	2000.0			

Mass Estimate Breakdown						
(kg) %						
Parametric	139.9	5.0%				
Calculated	1062.6	38.2%				
Measured	1575.9	56.7%				
Total	2778.5	100%				





#### **LAT Power Status**

1-Nov-04	Estimate	PARA	CALC	MEAS	ALLOC.
Item	(Watts)	(Watts)	(Watts)	(Watts)	(Watts)
ACD	11.5	2.4	3.9	5.2	10.5
Tracker	146.9	1.5	0.0	145.4	153.0
Calorimeter	66.8	0.0	0.0	66.8	65.0
Trigger & Data Flow	320.1	43.2	86.1	190.8	327.5
Grid/thermal	20.4	20.4	0.0	0.0	35.0
Instrument Total	565.6	67.5	90.0	408.2	591.0

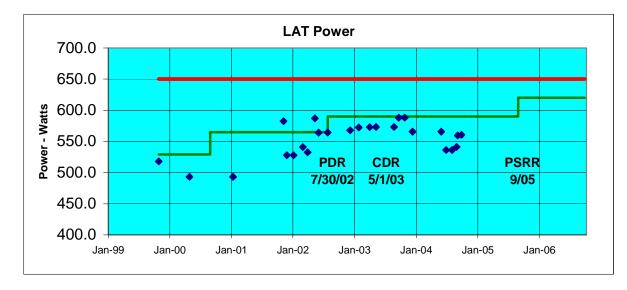
PDR Reserve Was 15.2%
CDR Reserve Was 13.4%
Goal for PSRR Reserve > 5%

Instrument Allocation 650.0
% Reserve 14.9%

PARA - Best Estimate based on conceptual design parameters
CALC - Estimate based on Calculated power from detailed design documentation

<u>MEAS</u> - Actual power measurements of components

Goals estimated using guidelines given in ANSI/AIAA G-020-1992 "Estimating and Budgeting Weight and Power Contingencies for Space Craft Systems"





### **LAT Power Status (Continued)**

#### Survival Power

Component	Current	Subsystem Power Estimates (W)				
	Alloc.	PARA	CALC	MEAS	Total	Margin
On-Orbit Average Power Total1	278.00	0.00	230.40	0.00	230.40	20.7%
Regulated VCHP Power Total	58.00	0.00	48.40	0.00	48.40	19.8%
Unregulated Passive Survival Power	220.00	0.00	182.00	0.00	182.00	20.9%

<sup>&</sup>lt;sup>1</sup>Power estimates reflect the LAT steady state orbit average. Numbers do not reflect transition into or out of survival mode, i.e. early orbit operations.



### **FSW Resource Usage Current Estimates**

Resource	Total Available	Current Usage	Margin Factor
EPU Boot PROM	256 kB	128 kB	2
SIU Boot PROM	256 kB	128 kB	2
EPU EEPROM	6 MB	1.5 MB	4
SIU EEPROM	6 MB	1.5-2.5 MB	3
EPU CPU cycles	200% in 2 EPUs	30%	> 6
SIU CPU cycles	100% in 1 SIU	25%	4
EPU memory	128 MB	16-32 MB	4-8
SIU memory	128 MB	< 16 MB	8



### **Instrument Bandwidth Resources**

#### • LAT communication, bandwidth (BW) in Mbyte/sec

Resource	Max Total BW limited by Hardware	Max limited by SC- ground transmissi on	Ave current BW at 10 KHz max trigger rate*	Ave current BW at 2 KHz nominal trigger rate*	Margin Factor (for 10 KHz rate)
Detector to GASU-EBM	45	N/A	10	2	4.5
GASU-EBM to EPU-CPU	20	N/A	5	1	4
EPU-CPU to GASU-EBM	2.5	0.075	0.04*	0.02*	2
GASU-EBM to SIU-CPU	5	0.15	0.08*	0.015*	2
SIU-CPU to Spacecraft	5	0.15	0.08*	0.015*	2

EBM: Event-Builder Module
EPU: Event-Processing Unit
SIU: Spacecraft Interface Unit

<sup>\*</sup> Present performance of event filter for EPU-CPU, still being optimized. Eventually the physics filter will be adjusted/loosened to take advantage of the max average bandwidh



### **Key Science Performance Metrics**

Parameter	SRD Value	Present Design Value
Peak Effective Area (in range 1-10 GeV)	>8000 cm <sup>2</sup>	10,000 cm <sup>2</sup> at 10 GeV
Energy Resolution 100 MeV on-axis	<10%	9%
Energy Resolution 10 GeV on-axis	<10%	8%
Energy Resolution 10-300 GeV on-axis	<20%	<15%
Energy Resolution 10-300 GeV off-axis (>60°)	<6%	<4.5%
PSF 68% 100 MeV on-axis	<3.5°	3.37° (front), 4.64° (total)
PSF 68% 10 GeV on-axis	<0.15°	0.086° (front), 0.115° (total)
PSF 95/68 ratio	<3	2.1 front, 2.6 back (100 MeV)
PSF 55°/normal ratio	<1.7	1.6
Field of View	>2sr	2.4 sr
Background rejection (E>100 MeV)	<10% diffuse	6% diffuse (adjustable)
Point Source Sensitivity(>100MeV)	<6x10 <sup>-9</sup> cm <sup>-2</sup> s <sup>-1</sup>	3x10 <sup>-9</sup> cm <sup>-2</sup> s <sup>-1</sup>
Source Location Determination	<0.5 arcmin	<0.4 arcmin (ignoring BACK info)
GRB localization	<10 arcmin	5 arcmin (ignoring BACK info)



## **Risk Management**



### **Risk Management Activity**

No new risks identified this month



## **Top risks**

ID#	Risk Rank	Risk Description	Risk Mitigation	Status
Proj Mgt - 003	Moderate	If completion of Tracker subsystem qualification is delayed due to EM or MCM electronics closure; then start of LAT I & T and schedule will be impacted	<ul> <li>Manufacturing Eng assigned to close MCM issues</li> <li>Increased team integration with Italian partners</li> <li>GSFC audit/support to Tracker EM closure</li> <li>Consider second source for tray and flex cable production</li> </ul>	<ul> <li>Restructured SLAC engineering support</li> <li>Additional INFN support in place</li> <li>Have 3 proposals for trays, downselect in process.</li> <li>Identified second source (Titan), development cables in work. Evaluating design mod to simplify production</li> </ul>
Proj Mgt - 002	Moderate	If ASICs fail to meet qualification requirements; then schedule impact will occur	Focused review & test. Margin for re-runs protected where possible     Individual risks Identified by subsystem     Extensive use of DAQ test bed to drive out system issues	•Cal/ACD ASIC's continued testing •Test Bed operating •No new issues
Proj Mgt - 004	Moderate	If TEM Power supply fails qualification; then final implementation may exceed schedule impacting delivery to I&T	*Key focus item identified for DAQ     *TEM/PS extensive EM use as EGSE	<ul> <li>Implementation plan in place and proceeding</li> <li>Fuse audit completed</li> <li>First article flight boards early November</li> </ul>



## **Top risks**

ID#	Risk Rank	Risk Description	Risk Mitigation	Status
SE-007	Moderate	If a critical component fails post LAT integration; then de- integration will result in cost & schedule impact	Extensive use of EM test bed to support flight H/W & S/W development     Thorough qualification and acceptance tests     Pre planned I&T actions for deintegration	<ul> <li>Qual &amp; acceptance planning in-place</li> <li>I&amp;T developing rework contingency plans.</li> <li>Integration plan baselined</li> </ul>
Elec- 004	Moderate	If target hardware, requirement development or manpower is delayed; Then Flight-Software development schedule will be impacted	<ul> <li>Detailed incremental development program</li> <li>Ensure sufficient software test on target hardware during development to drive out any requirement disconnects.</li> <li>Include adequate peer reviews before each spiral cycle prior to release</li> <li>Include monthly Demos to verify functionality/measure progress</li> </ul>	•Adapting monthly demos •Tracking EGSE resource utilization •Hired FSW manager •Successful FSW review on 16 September •Continuing monthly demos •Updated detailed test plan in release cycle incorporating dedicated test team



# **Top risks**

ID#	Risk Rank	Risk Description	Risk Mitigation	Status
Proj Mgt - 005	Moderate	If parts and vendor orders are delayed or bids exceed expectations; then flight production costs & delivery schedule will be impacted	Manufacturing engineer added to expedite minimum cost closure     Clarification and purchase package review to ensure accurate bids     Increase production management staff	<ul> <li>Purchase order tracking/monitoring system in place to highlight roadblocks</li> <li>Design documentation release plan prioritized by vendor selection and component fabrication need dates</li> <li>Workarounds implemented for late parts</li> <li>Hired additional head to manage production</li> </ul>
IT - 006	Moderate	If logistic or facility integration issues are found during LAT environmental test program; then re-work will delay schedule	LAT I&T to plan a roadmap of activities from LAT building 33 to completion of environmental testing      LAT I&T to consider and develop opportunities to path find key activities required prior to LAT shipment to NRL	•Follow up Environmental Planning TIM held on 1 October at SLAC, I&T driving Als to conclusion