



### **GLAST Large Area Telescope:**

**LAT System Engineering** 

Pat Hascall SLAC System Engineering



### **Topics**

- Action Item Status
- Technical Baseline Management
- Issues
- Interface Control Documentation
- RFA Closure
- Key Metrics
- Risk Management



### **Monthly Action Item Status**

Action Item ID	Actionee	Description	Status
7-30-03-008	B. Estey	Define and maintain the production readiness/execution plan to include vendor selection and associated schedule to ensure unit availability dates are met	OPEN: Draft production plan completed & provided to GSFC. Refinement required as vendors are selected. Update provided early December, 2003. Next update and process for update: TBD. Schedules for TEM/TPS provided to B.Graf, action to be closed when similar schedules are provided for the rest of the boxes. SIU/EPU expected Jan 19->28, Heater Control Box expected Jan 26->Feb 9.  Meeting scheduled with Bernie, Brigitte and Pat for Mar 2



# Technical Baseline: Flight Drawing Release

- Status details (DAQ reported separately)
  - Tracker
    - 141 of 141 completed (total is 15 over original plan)
  - ACD
    - Last assembly drawing in signoff
  - Mech
    - Completed 61 of 81 (total is 22 part over original plan)
      - 16 MLI drawings will be reduced based on successful meeting with Spectrum to simplify blanket design and interface
      - Remaining 4 drawings (shims and spacers) are needed in July
  - Design Integration
    - Major drawings: 2 of 6 signed off



### **Technical Baseline: DAQ Flight Drawing Release**

		In Config	То	In Sign	
Group	Total	Control	Go	off	Notes
TEM/TPS	48	48	0		
PDU	34	34	0		
GASU	69	56	13		13 to close with FPGA docs*
EPU/SIU	59	56	3	3	3 FPGA documents have started signoff process*
Harness	35	35	0		
					2 brackets coupled to MLI resolution, need date is post
Brackets/hardware	35	33	2		tower integration
Heater Control Box	21	1	20		

<sup>\*</sup> FPGA document release held to allow more checkout time before FPGA program burn in



### Issues

No.	Description	Status	Due Date	Actionee
3	Technical baseline:	-All drawings to be under CM prior	Weekly Review	P. Hascall
	Flight Drawing	to flight build		
	release	-Flight drawing release plan		
		generated and statused weekly		
22	ASIC radiation	Radiation testing scheduled for	30 April ->June-	Sadrozinksi
	testing status	completion. GLTC TID in March	>Jan 05 >March 05	
24	No plans to conduct	Looking at an EMI/EMC test to be	30 Sept>March	Himel
	Tracker Subsystem	performed after Tracker delivery		
	EMI/EMC	but before integration. Tracker B		
		AT complete, Tracker 1 qual test		
		end of March		



# **Issues (continued)**

No.	Description	Status	Due Date	Actionee
31	Tracker flex cable	Process change implemented. Coupons from	10/15/04-	Rich
	coupon failures	flight panels failed. Steve Kahn assigned to	>11/5> 1/31	
		work with Parlex on quality and schedule.		
32	Tracker wire bond	Evaluating root cause. Potentially delete		R. Johnson
	breaks (heavy	encapsulation. Tower A and B to proceed		
	trays) pitch	with encapsulation. Tower 1 and following		
	adapter to ladder	will not have encapsulation		
35	Reliability	FMEAs done, reviews with Subsystems	12/31/04	DiVenti
	assessments not	started. Held TKR and Mech reviews with		
	completed	SLAC, ELX review potentially on Friday		
36	SIIS capability to	SLAC will build EGSE to support I&T	Closed	Haller/
	support I&T			Bloom
37	SIB EEPROM	PCB approved enough parts for flight		Haller
	DPA Failure	build, still working parts for spares and		
		qual		



# **Issues (continued)**

No.	Description	Status	Due Date	Actionee
38	RAD750 heat sink	Heat sinks reworked and Omnirel		Haller
	and Omnirel alert	regulators to be replaced.		
39	LAT Stay-Clear	Successful series of meetings, change paper	31 Jan	Bielawski
	Violations	underway		
40	LAT-DAQ FPGA	SLAC to respond to AIs' from FPGA reviews		Haller
	development and	– AI's in review		
	qualification			
41	Qualification of	BAE to respond to GSFC solder joint		Haller
	ERNI connectors	analysis		
		Trial runs for LAT ERNI connectors at		
		Aeroflex		
42	Pitch Adapter to	Will use screened parts through tower 3.		Johnson
	GTFE bond	Validation process underway for improved		
	breaks	encapsulation. Covers as a backup		



### **Interface Management**



### **Interface Document Status**

- SC-LAT ICD ICN Status
  - LAT signed this month
    - None
  - Currently under signature review
    - None
  - Currently in draft or revision
    - None
- Internal LAT ICD's
  - Signed Off
    - None
  - Currently in signature review
    - None
  - Currently in update
    - Electronics-LAT ICD (Comments being incorporated as they are received)
    - CAL-LAT ICD



### Deliverables/Receivables

#### LAT Deliverables

Feb: None Scheduled

Mar: ISIS Training?? Spectrum unable to provide firm date.

Apr: None Scheduled

May: None Scheduled

#### LAT Receivables

– Feb: None

– Mar: SIIS

Apr: None Scheduled

May: None Scheduled



# **Key Design Metrics**



### **Mass and Power Status Summary**

#### Mass

- Have measured values for Calorimeters, two Trackers, and one TEM/TPS
- No significant issues (within a few percent of predicts)
- Will update the mass report when the grid is measured

#### Power

- Updated survival power budget
  - Supports higher Survival Cold temperatures to minimize Tracker cold exposure
  - Also corrected typo, so the net result was an improvement in margin
- Measured data for Tower A indicates a potential increase in LAT power estimates of around 20 watts



### **LAT Mass Status**

LAT Mass Status Report

LAT-TD-00564-10

LAT Mass Status Martin Nordby Effective Date: 15-Sep-04

Print Date: 15-Sep-04

Sep-04

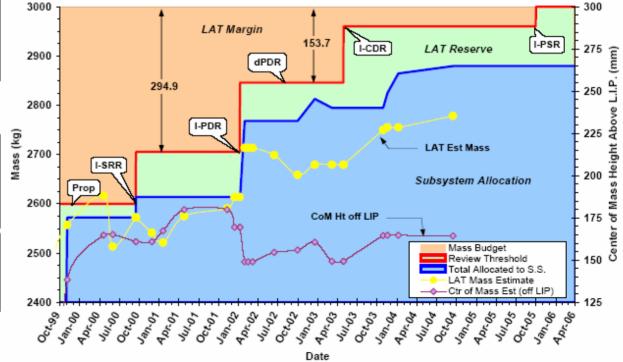
Mass (kg)	Estimate	Alloc.
TKR	514.0	510.0
CAL	1374.3	1440.0
ACD	286.2	295.0
Mech	366.6	386.6
Elec	230.4	240.0
Systems	7.0	8.0
LAT Total	2778.5	2879.6
Rsrv/Margin	221.5	
Rsrv/Margin*	8.0%	
Allocation		3000.0

<sup>\*</sup> AIAA G-020 recommended min reserve = 5.2% Allocations per latest mass CCB on 18 June 2004

Center of Mass (mm)					
CMx	-1.22	-20 < CMx < 20			
CMy	-0.89	-20 < CMy < 20			
CMz	-72.55	CMz < -51.2			
Ht off LIP	163.65	Ht < 185			

Second Moment of Inertia (kg-m²)					
lxx	1084.5	1500.0			
lyy	1032.1	1500.0			
Izz	1410.8	2000.0			

Mass Estimate Breakdown							
(kg) %							
Parametric	139.9	5.0%					
Calculated	1062.6	38.2%					
Measured 1575.9 56.7%							
Total 2778.5 100%							





### **LAT Power Status**

Calorimeter CR approved to change allocation to 71W

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1-Nov-04	Estimate	PARA	CALC	MEAS	ALLOC.
Item	(Watts)	(Watts)	(Watts)	(Watts)	(Watts)
ACD	11.5	2.4	3.9	5.2	10.5
Tracker	146.9	1.5	0.0	145.4	153.0
Calorimeter	66.8	0.0	0.0	66.8	65.0
Trigger & Data Flow	320.1	43.2	86.1	190.8	327.5
Grid/thermal	20.4	20.4	0.0	0.0	35.0
Instrument Total	565.6	67.5	90.0	408.2	591.0

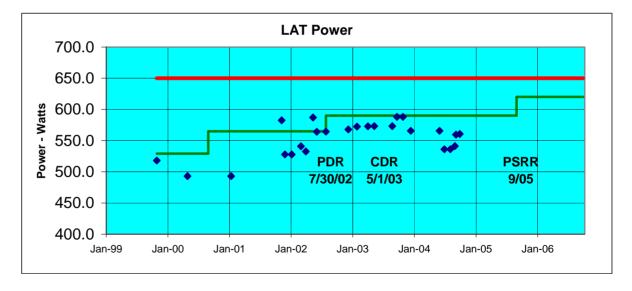
PDR Reserve Was 15.2% CDR Reserve Was 13.4%

Goal for PSRR Reserve > 5%

Instrument Allocation 650.0
% Reserve 14.9%

PARA - Best Estimate based on conceptual design parameters
CALC - Estimate based on Calculated power from detailed design documentation
MEAS - Actual power measurements of components

Goals estimated using guidelines given in ANSI/AIAA G-020-1992 "Estimating and Budgeting Weight and Power Contingencies for Space Craft Systems"





### **LAT Power Status (Continued)**

#### Survival Power

Component	Current	st Subsystem Power Estimates (W)			(W)	
	Alloc.	PARA	CALC	MEAS	Total	Margin
On-Orbit Average Power Total <sup>1</sup>	278.00	0.00	203.00	0.00	203.00	36.90%
Regulated VCHP Power Total	58.00	0.00	43.00	0.00	43.00	34.90%
Unregulated Passive Survival Power	220.00	0.00	160.00	0.00	160.00	37.50%

<sup>&</sup>lt;sup>1</sup>Power estimates reflect the LAT steady state orbit average. Numbers do not reflect transition into or out of survival mode, i.e. early orbit operations.



# **FSW Resource Usage Current Estimates**

Resource	Total Available	Current Usage	Margin Factor
EPU Boot PROM	256 kB	128 kB	2
SIU Boot PROM	256 kB	128 kB	2
EPU EEPROM	6 MB	1.5 MB	4
SIU EEPROM	6 MB	1.5-2.5 MB	3
EPU CPU cycles	200% in 2 EPUs	30%	> 6
SIU CPU cycles	100% in 1 SIU	25%	4
EPU memory	128 MB	16-32 MB	4-8
SIU memory	128 MB	< 16 MB	8



### **Instrument Bandwidth Resources**

#### • LAT communication, bandwidth (BW) in Mbyte/sec

Resource	Max Total BW limited by Hardware	Max limited by SC- ground transmissi on	Ave current BW at 10 KHz max trigger rate*	Ave current BW at 2 KHz nominal trigger rate*	Margin Factor (for 10 KHz rate)
Detector to GASU-EBM	45	N/A	10	2	4.5
GASU-EBM to EPU-CPU	20	N/A	5	1	4
EPU-CPU to GASU-EBM	2.5	0.075	0.04*	0.02*	2
GASU-EBM to SIU-CPU	5	0.15	0.08*	0.015*	2
SIU-CPU to Spacecraft	5	0.15	0.08*	0.015*	2

EBM: Event-Builder Module
EPU: Event-Processing Unit
SIU: Spacecraft Interface Unit

<sup>\*</sup> Present performance of event filter for EPU-CPU, still being optimized. Eventually the physics filter will be adjusted/loosened to take advantage of the max average bandwidh



### **Key Science Performance Metrics**

Parameter	SRD Value	Present Design Value
Peak Effective Area (in range 1-10 GeV)	>8000 cm <sup>2</sup>	10,000 cm <sup>2</sup> at 10 GeV
Energy Resolution 100 MeV on-axis	<10%	9%
Energy Resolution 10 GeV on-axis	<10%	8%
Energy Resolution 10-300 GeV on-axis	<20%	<15%
Energy Resolution 10-300 GeV off-axis (>60°)	<6%	<4.5%
PSF 68% 100 MeV on-axis	<3.5°	3.37° (front), 4.64° (total)
PSF 68% 10 GeV on-axis	<0.15°	0.086° (front), 0.115° (total)
PSF 95/68 ratio	<3	2.1 front, 2.6 back (100 MeV)
PSF 55°/normal ratio	<1.7	1.6
Field of View	>2sr	2.4 sr
Background rejection (E>100 MeV)	<10% diffuse	6% diffuse (adjustable)
Point Source Sensitivity(>100MeV)	<6x10 <sup>-9</sup> cm <sup>-2</sup> s <sup>-1</sup>	3x10 <sup>-9</sup> cm <sup>-2</sup> s <sup>-1</sup>
Source Location Determination	<0.5 arcmin	<0.4 arcmin (ignoring BACK info)
GRB localization	<10 arcmin	5 arcmin (ignoring BACK info)



# **Risk Management**



### **Risk Management Activity**

No changes



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
Proj Mgt - 002	Moderate	If ASICs fail to meet qualification requirements; then schedule impact will occur	Focused review & test. Margin for re-runs protected where possible     Individual risks Identified by subsystem     Extensive use of DAQ test bed to drive out system issues	Cal/ACD ASIC's continued testing     Test Bed operating     No new issues
Proj Mgt - 004	Moderate	If TEM Power supply fails qualification; then final implementation may exceed schedule impacting delivery to I&T	*Key focus item identified for DAQ     *TEM/PS extensive EM use as     EGSE	<ul> <li>Implementation plan in place and proceeding</li> <li>Fuse audit completed</li> <li>Functional tests complete, environmental tests to be completed in February</li> </ul>



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
SE-007	Moderate	If a critical component fails post LAT integration; then de- integration will result in cost & schedule impact	Extensive use of EM test bed to support flight H/W & S/W development     Thorough qualification and acceptance tests     Pre planned I&T actions for deintegration	<ul> <li>Qual &amp; acceptance planning in-place</li> <li>I&amp;T developing rework contingency plans.</li> <li>Integration plan baselined</li> </ul>
Elec- 004	Moderate	If target hardware, requirement development or manpower is delayed; Then Flight-Software development schedule will be impacted	<ul> <li>Detailed incremental development program</li> <li>Ensure sufficient software test on target hardware during development to drive out any requirement disconnects.</li> <li>Include adequate peer reviews before each spiral cycle prior to release</li> <li>Include monthly Demos to verify functionality/measure progress</li> </ul>	•Adapting monthly demos •Tracking EGSE resource utilization •Updated detailed test plan released •Demo frequency increased from monthly to approximately weekly



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
Proj Mgt - 005	Moderate	If parts and vendor orders are delayed or bids exceed expectations; then flight production costs & delivery schedule will be impacted	Manufacturing engineer added to expedite minimum cost closure     Clarification and purchase package review to ensure accurate bids     Increase production management staff	Purchase order tracking/monitoring system in place to highlight roadblocks  Design documentation release plan prioritized by vendor selection and component fabrication need dates  Workarounds implemented for late parts  Hired additional head to manage production
IT - 006	Moderate	If logistic or facility integration issues are found during LAT environmental test program; then re-work will delay schedule	LAT I&T to plan a roadmap of activities from LAT building 33 to completion of environmental testing      LAT I&T to consider and develop opportunities to path find key activities required prior to LAT shipment to NRL	Follow up Environmental     Planning TIM held on 1     October at SLAC, I&T driving     Als to conclusion     Continuing periodic TIMS



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
SE - 011	Low	If individual tracker towers do not meet performance requirements due to manufacturing issues (e.g. wire bond breaks) then the LAT may not meet science requirements	Understand stability of performance to determine mitigation strategies Limit LAT temperature excursions to minimize possible propagation of some types of tracker issues  Optimize placement of towers based on individual tower performance to minimize science effects and to minimize removal and replacement efforts should they become necessary	Temperature range reduced at the LAT level to allow a narrower range during Tracker and LAT tests  Alternate plan for placement of Tracker A and B being implemented