



GLAST Large Area Telescope:

LAT System Engineering

Pat Hascall SLAC System Engineering



Topics

- Action Item Status
- Technical Baseline Management
- Issues
- Interface Control Documentation
- RFA Closure
- Key Metrics
- Risk Management



Monthly Action Item Status

Action Item ID	Actionee	Description	Status
7-30-03-008	B. Estey	Define and maintain the production readiness/execution plan to include vendor selection and associated schedule to ensure unit availability dates are met	OPEN: Draft production plan completed & provided to GSFC. Refinement required as vendors are selected. Update provided early December, 2003. Next update and process for update: TBD. Schedules for TEM/TPS provided to B.Graf, action to be closed when similar schedules are provided for the rest of the boxes. SIU/EPU expected Jan 19->28, Heater Control Box expected Jan 26->Feb 9. Telecon with Bernie and Brigitte resulted in a milestone format chart. First delivery provided, action closed.



Technical Baseline: Flight Drawing Release

- Status details (DAQ reported separately)
 - Tracker
 - 141 of 141 completed (total is 15 over original plan)
 - ACD
 - Completed
 - Mech
 - Completed 63 of 79 (total is 20 over original plan)
 - 12 MLI drawings in process, several are entering the review cycle
 - Remaining 4 drawings (shims and spacers) are needed in July
 - Design Integration
 - Major drawings: 2 of 6 signed off



Technical Baseline: DAQ Flight Drawing Release

Group	Total	In Config Control	To Go	In Sign off	Notes
TEM/TPS	48	48	0		
PDU	34	34	0		
GASU	69	69	0		
EPU/SIU	59	59	0		
Harness	35	35	0		
Brackets/hardware	35	33	2		2 brackets coupled to MLI resolution, need date is post tower integration
Heater Control Box	20	20	0		One drawing deleted, was 21



Issues

No.	Description	Status	Due Date	Actionee
3	Technical baseline:	-All drawings to be under CM prior	Weekly Review	P. Hascall
	Flight Drawing	to flight build		
	release	-Flight drawing release plan		
		generated and statused weekly		
22	ASIC radiation	Radiation testing scheduled for	30 April ->June-	Sadrozinksi
	testing status	completion. GLTC tests	>Jan 05 >March 05	
		successfully completed,		
		documentation in work		
24	No plans to conduct	Looking at an EMI/EMC test to be	30 Sept>March	Himel
	Tracker Subsystem	performed after Tracker delivery		
	EMI/EMC	but before integration. Tracker B		
		AT complete, Tracker 2 qual test		
		complete, one waiver in work		



Issues (continued)

No.	Description	Status	Due Date	Actionee
31	Tracker flex cable	Process change implemented. Coupons from	10/15/04-	Rich
	coupon failures	flight panels failed. Steve Kahn assigned to	>11/5> 1/31	
		work with Parlex on quality and schedule. PO		
		in place to Pioneer as a second source		
35	Reliability	FMEAs done, reviews with Subsystems	12/31/04	DiVenti
	assessments not	started. Held TKR and Mech reviews with		
	completed	SLAC, ELX review on 4/22 covered EPU,		
		SIU and GASU. TPS, GASU and PDU		
		tentatively scheduled for 4/29		
37	SIB EEPROM	PCB approved enough parts for flight build,		Haller
	DPA Failure	still working parts for spares and qual		



Issues (continued)

No.	Description	Status	Due Date	Actionee
38	RAD750 heat sink	Heat sinks reworked and Omnirel regulators		Haller
	and Omnirel alert	to be replaced.		
39	LAT Stay-Clear	Successful series of meetings, change paper	31 Jan	Bielawski
	Violations	underway. GSFC action to coordinate		
		GBM thermal field of view analysis.		
40	LAT-DAQ FPGA	SLAC to respond to AIs' from FPGA reviews		Haller
	development and	– AI's in review		
	qualification			
41	Qualification of	BAE to respond to GSFC solder joint analysis		Haller
	ERNI connectors	Trial runs for LAT ERNI connectors at		
		Aeroflex had issues, process in work		
43	TEM/TPS voltage	Combination of hardware test and modeling		Haller
	ripple	in process to determine cause and potential		
		fixes. Cause and corrective action		
		determined, retrofit in process		
44	Tower 1 ladder	Tray replaced, Tower 1 going through	Closed	Johnson
	intermittent	regression testing. Testing completed		
		successfully. Testing on removed tray to		
		validate failure mode theory. Failure		
		scenario verified.		
45	ACD PMT Noise	Several ACD channels showed noise		D.
		during high bias voltage tests.		Thompson



Interface Management



Interface Document Status

- F2F TIM @ Spectrum scheduled for the week of May 16 to close remaining ICD issues and to discuss Observatory I&T activities
- SC-LAT ICD ICN Status
 - LAT signed this month
 - None
 - Currently under signature review
 - None
 - Currently in draft or revision
 - ICN-087 LAT Deliveries Table
 - ICN-090 LAT Current Transients
 - ICN-0XX LAT Survey Feature Locations and Access Requirements
- Internal LAT ICD's
 - Signed off this month
 - None
 - Currently in signature review
 - None
 - Currently in update
 - CAL-LAT ICD
 - CR and DCN written and will be submitted for signature review this week
 - ACD-LAT ICD
 - CR and DCN written, working final issue before sending out for signature review
 - Electronics-LAT ICD (Comments being incorporated as they are received)



Deliverables/Receivables

LAT Deliverables

Apr: None Scheduled

May: None Scheduled

June: None Scheduled

July: None Scheduled

LAT Receivables

– Apr: None

May: SIIS and SIIS harness

June: None Scheduled

July: None Scheduled



Key Design Metrics



Mass and Power Status Summary

Mass

- Formal update in process
- No significant issues on measured data (within a few percent of predicts)

Power

- No change to budget
- Potential increase based on Tower A and Tower B measurement was estimated to be 13.8 W for the complete 16 towers, but want to see a few more towers before the LAT budget is updated



LAT Mass Status

LAT Mass Status Report LAT-TD-00564-10

LAT Mass Status

Martin Nordby

LAT-TD-00564-10

Effective Date: 15-Sep-04

Print Date: 15-Sep-04

Sep-04

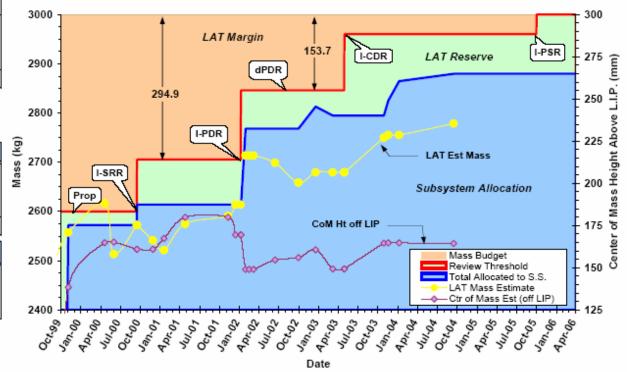
Mass (kg)	Estimate	Alloc.
TKR	514.0	510.0
CAL	1374.3	1440.0
ACD	286.2	295.0
Mech	366.6	386.6
Elec	230.4	240.0
Systems	7.0	8.0
LAT Total	2778.5	2879.6
Rsrv/Margin	221.5	
Rsrv/Margin*	8.0%	
Allocation		3000.0

^{*} AIAA G-020 recommended min reserve = 5.2% Allocations per latest mass CCB on 18 June 2004

Center of Mass (mm)					
CMx	-1.22	-20 < CMx < 20			
CMy	-0.89	-20 < CMy < 20			
CMz	-72.55	CMz < -51.2			
Ht off LIP	163.65	Ht < 185			

Second Moment of Inertia (kg-m²)					
lxx	1084.5	1500.0			
lyy	1032.1	1500.0			
Izz	1410.8	2000.0			

Mass Estimate Breakdown						
	(kg)	%				
Parametric	139.9	5.0%				
Calculated	1062.6	38.2%				
Measured	1575.9	56.7%				
Total	2778.5	100%				





LAT Power Status

30-Mar-05	Estimate	PARA	CALC	MEAS	ALLOC.
Item	(Watts)	(Watts)	(Watts)	(Watts)	(Watts)
ACD	11.5	2.4	3.9	5.2	10.5
Tracker	146.9	1.5	0.0	145.4	153.0
Calorimeter	67.4	0.0	0.6	66.8	71.0
Trigger & Data Flow	321.2	43.2	87.2	190.8	327.5
Grid/thermal	20.4	20.4	0.0	0.0	35.0
Instrument Total	567.4	67.5	91.7	408.2	597.0
				•	

Instrument Allocation 650.0

% Reserve 14.6%

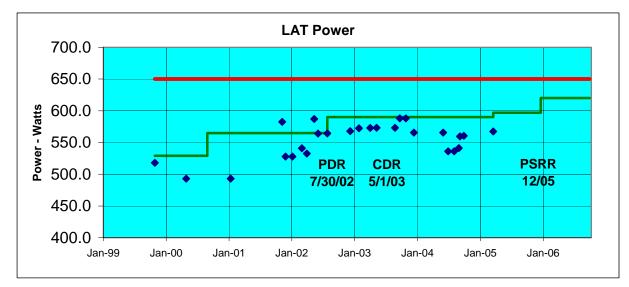
PARA - Best Estimate based on conceptual design parameters

CALC - Estimate based on Calculated power from detailed design documentation

<u>MEAS</u> - Actual power measurements of components

Goals estimated using guidelines given in ANSI/AIAA G-020-1992 "Estimating and Budgeting Weight and Power Contingencies for Space Craft Systems"







Measured LAT Power

Tower Assembly Power Consumption						
Tower Estimate	wer Estimate 25.93 Watts					
			TKR w/bias and			
Tower	Bay	Power	CAL	TEM/TPS	Calorimeter	Tracker
Α		26.86	0.956	1752	FM104	Α
В		26.72	0.951	1753	FM105	В
1						
2						
3						
4						
5						
6						
7						
8						
9						
10						
11						
12						
13						
14						
Tower Average		26.79				•
LAT Estimate		428.69				
LAT Increase		13.81				

- 1. Tower A and B measurements consistent and above predicts
- 2. TEM/TPS measured values are based on telemetry and are relatively inaccurate

TEM/TPS Assembly Power Consumption					
Estimate		12.54	Watts		
Serial #	Cold	Ambient	Hot		
GLAT1754[1]					
GLAT1752	8.97	10.73	10.36		
GLAT1753	11.25	11.16	13.35		
GLAT1755					
GLAT1756					
GLAT1757					
GLAT1758					
GLAT1759					
GLAT1760					
GLAT1761					
GLAT1762					
GLAT1763					
GLAT1764					
GLAT1765					
GLAT1766					
GLAT1767					
GLAT1768					
Average	10.11	10.95	11.86		
LAT Estimate	161.69	175.14	189.71		
[1]	Qualificatio	n Unit			

Calorimeter Power Consumption							
Estimate		4.21	Watts				
Serial #	-30	25	50				
FM101 [1]	4.48	4.12	3.96				
FM102	4.59	4.22	4.06				
FM103	4.60	4.24	4.07				
FM104	4.59	4.25	4.07				
FM105	4.55	4.22	4.02				
FM106	4.61	4.21	4.05				
FM107	4.53	4.25	3.99				
FM108							
FM109							
FM110							
FM111							
FM112							
FM113							
FM114							
FM115							
FM116							
Average	4.56	4.22	4.03				
LAT Estimate	73.02	67.45	64.51				
[1]	Proto Flight	t Unit					

Estimate		9.18	Watts					
Serial #	Cold	Ambient	Hot					
A [1]		9.70						
В		9.80						
1								
2		9.96						
3								
4								
5								
6								
7								
8								
9								
10								
11								
12								
13								
14								
Average	0.00	9.82	0.00					
LAT Estimate	0.00	157.12	0.00					

Tracker Module Power Consumption



LAT Power Status (Continued)

Survival Power

Component	Current	Sı	ubsystem	Power E	stimates	(W)
	Alloc.	PARA	CALC	MEAS	Total	Margin
On-Orbit Average Power Total ¹	278.00	0.00	203.00	0.00	203.00	36.90%
Regulated VCHP Power Total	58.00	0.00	43.00	0.00	43.00	34.90%
Unregulated Passive Survival Power	220.00	0.00	160.00	0.00	160.00	37.50%

¹Power estimates reflect the LAT steady state orbit average. Numbers do not reflect transition into or out of survival mode, i.e. early orbit operations.





FSW Resource Usage Current Estimates

Resource	Total Available	Current Usage	Margin Factor
EPU Boot PROM	256 kB	128 kB	2
SIU Boot PROM	256 kB	128 kB	2
EPU EEPROM	6 MB	1.5 MB	4
SIU EEPROM	6 MB	1.5-2.5 MB	3
EPU CPU cycles	200% in 2 EPUs	30%	> 6
SIU CPU cycles	100% in 1 SIU	25%	4
EPU memory	128 MB	16-32 MB	4-8
SIU memory	128 MB	< 16 MB	8



Instrument Bandwidth Resources

• LAT communication, bandwidth (BW) in Mbyte/sec

Resource	Max Total BW limited by Hardware	Max limited by SC- ground transmissi on	Ave current BW at 10 KHz max trigger rate*	Ave current BW at 2 KHz nominal trigger rate*	Margin Factor (for 10 KHz rate)
Detector to GASU-EBM	45	N/A	10	2	4.5
GASU-EBM to EPU-CPU	20	N/A	5	1	4
EPU-CPU to GASU-EBM	2.5	0.075	0.04*	0.02*	2
GASU-EBM to SIU-CPU	5	0.15	0.08*	0.015*	2
SIU-CPU to Spacecraft	5	0.15	0.08*	0.015*	2

EBM: Event-Builder Module
EPU: Event-Processing Unit
SIU: Spacecraft Interface Unit

^{*} Present performance of event filter for EPU-CPU, still being optimized. Eventually the physics filter will be adjusted/loosened to take advantage of the max average bandwidh



Key Science Performance Metrics

Parameter	SRD Value	Present Design Value
Peak Effective Area (in range 1-10 GeV)	>8000 cm ²	10,000 cm ² at 10 GeV
Energy Resolution 100 MeV on-axis	<10%	9%
Energy Resolution 10 GeV on-axis	<10%	8%
Energy Resolution 10-300 GeV on-axis	<20%	<15%
Energy Resolution 10-300 GeV off-axis (>60°)	<6%	<4.5%
PSF 68% 100 MeV on-axis	<3.5°	3.37° (front), 4.64° (total)
PSF 68% 10 GeV on-axis	<0.15°	0.086° (front), 0.115° (total)
PSF 95/68 ratio	<3	2.1 front, 2.6 back (100 MeV)
PSF 55°/normal ratio	<1.7	1.6
Field of View	>2sr	2.4 sr
Background rejection (E>100 MeV)	<10% diffuse	6% diffuse (adjustable)
Point Source Sensitivity(>100MeV)	<6x10 ⁻⁹ cm ⁻² s ⁻¹	3x10 ⁻⁹ cm ⁻² s ⁻¹
Source Location Determination	<0.5 arcmin	<0.4 arcmin (ignoring BACK info)
GRB localization	<10 arcmin	5 arcmin (ignoring BACK info)



Risk Management



Risk Management Activity

No changes



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
Proj Mgt - 002	Moderate	If ASICs fail to meet qualification requirements; then schedule impact will occur	•Focused review & test. Margin for re-runs protected where possible •Individual risks Identified by subsystem •Extensive use of DAQ test bed to drive out system issues	Cal/ACD ASIC's continued testing Test Bed operating No new issues
Proj Mgt - 004	Moderate	If TEM Power supply fails qualification; then final implementation may exceed schedule impacting delivery to I&T	*Key focus item identified for DAQ *TEM/PS extensive EM use as EGSE	 Implementation plan in place and proceeding Fuse audit completed Testing complete, data package in work



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
SE-007	Moderate	If a critical component fails post LAT integration; then de- integration will result in cost & schedule impact	Extensive use of EM test bed to support flight H/W & S/W development Thorough qualification and acceptance tests Pre planned I&T actions for deintegration	 Qual & acceptance planning in-place I&T developing rework contingency plans. Integration plan baselined
Elec- 004	Moderate	If target hardware, requirement development or manpower is delayed; Then Flight-Software development schedule will be impacted	Detailed incremental development program Ensure sufficient software test on target hardware during development to drive out any requirement disconnects. Include adequate peer reviews before each spiral cycle prior to release Include monthly Demos to verify functionality/measure progress	Adapting monthly demos Tracking EGSE resource utilization Updated detailed test plan released Demo frequency increased from monthly to approximately weekly



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
Proj Mgt - 005	Moderate	If parts and vendor orders are delayed or bids exceed expectations; then flight production costs & delivery schedule will be impacted	Manufacturing engineer added to expedite minimum cost closure Clarification and purchase package review to ensure accurate bids Increase production management staff	Purchase order tracking/monitoring system in place to highlight roadblocks Design documentation release plan prioritized by vendor selection and component fabrication need dates Workarounds implemented for late parts Hired additional head to
IT - 006	Moderate	If logistic or facility integration issues are found during LAT environmental test program; then re-work will delay schedule	LAT I&T to plan a roadmap of activities from LAT building 33 to completion of environmental testing LAT I&T to consider and develop opportunities to path find key activities required prior to LAT shipment to NRL	 manage production Follow up Environmental Planning TIM held on 1 October at SLAC, I&T driving Als to conclusion Continuing periodic TIMS, May 3-4 for the next TIM



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
SE - 011	Low	If individual tracker towers do not meet performance requirements due to manufacturing issues (e.g. wire bond breaks) then the LAT may not meet science requirements	Understand stability of performance to determine mitigation strategies Limit LAT temperature excursions to minimize possible propagation of some types of tracker issues Optimize placement of towers based on individual tower performance to minimize science effects and to minimize removal and replacement efforts should they become necessary	Temperature range reduced at the LAT level to allow a narrower range during Tracker and LAT tests Alternate plan for placement of Tracker A and B being implemented



Cost Report

Reporting Category	Cost Incurred		Estimated Cost			Estimated Final Cost		Unfilled Orders		
	During Month Cum. to Date		De	tail	Balance of	Contractor	Contract	Outstanding		
	Actual	Planned	Actual	Planned	APR05	MAY05	Contract	Estimate	Value	
4.1.2 SYSTEM ENGINEERING										
4.1.2.1 REQ'TS MGMT, DESIGN INTEGRATION & TEST	-31	48	3,007	3,086	27	27	191	3,253	3,253	0
4.1.2.3 SYSTEM ANALYSIS	12	11	1,027	970	11	10	-14	1,034	1,034	
4.1.2.4 QUALIFICATION & TRACKING	108	64	374	330	59	59	197	689	689	
4.1.2.5 RISK & RELIABILITY ANALYSIS			99	98			-1	98	98	
4.1.2.6 CONFIGURATION MGMT & DOCUMENT / DATA LIE	8	9	274	266	8	9	30	321	321	
4.1.2.7 MANAGEMENT & PLANNING	73	79	1,853	1,953	75	60	264	2,252	2,252	414
CAPW[3]Totals:	169	213	6,635	6,704	180	166	666	7,647	7,647	414

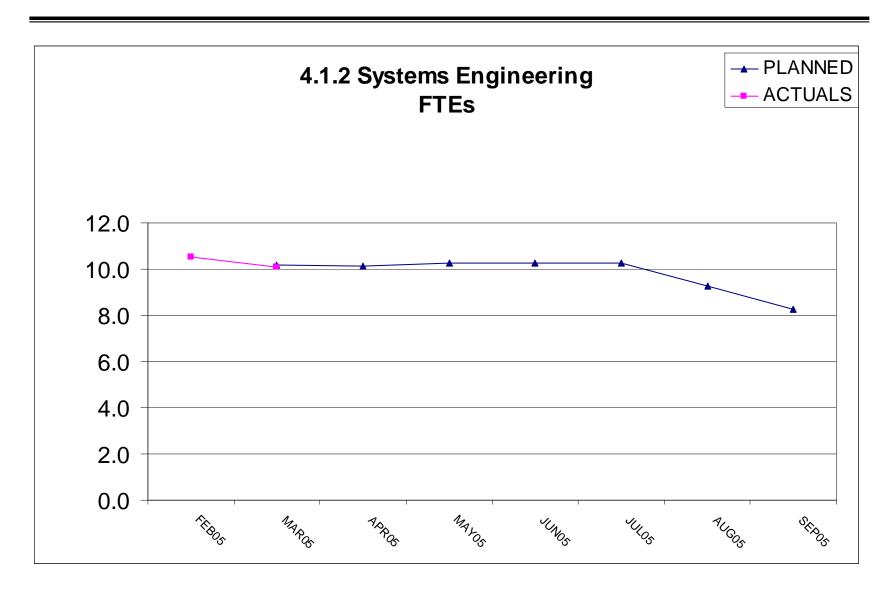


Cost Variance Explanation

- Why overrun/underrun?
- What will be done to correct?



FTE Report





FTE Variance Explanation

- Why overrun/underrun?
 - No overrun/underrun
- What is the impact?
 - No impact
- What will be done to correct?
 - No corrective action required