



GLAST Large Area Telescope:

LAT System Engineering

Pat Hascall SLAC System Engineering



Topics

- Action Item Status
- Technical Baseline Management
- Issues
- Interface Control Documentation
- RFA Closure
- Key Metrics
- Risk Management



Monthly Action Item Status

Action Item ID	Actionee	Description	Status



Technical Baseline: Flight Drawing Release

- Status details
 - Tracker
 - 141 of 141 released
 - ACD
 - Completed
 - Calorimeter
 - Completed
 - Mech
 - Completed 66 of 73
 - 3 MLI drawings in work
 - Remaining 4 drawings (shims and spacers) are needed in for radiator fit check (August earliest)
 - Design Integration
 - Major drawings: 4 of 7 signed off
 - Added External Cable Installation Kit
 - DAQ
 - 300 of 300 released



Issues

	Γ	Τ	T	T
No.	Description	Status	Due Date	Actionee
3	Technical	-All drawings to be under CM prior to flight	Weekly	P. Hascall
	baseline: Flight	build	Review	
	Drawing release	-Flight drawing release plan generated and		
		statused weekly		
22	ASIC radiation	Radiation testing scheduled for completion.	30 April -	Sadrozinksi
	testing status	Request to eliminate TID for 3 rd and 4 th	>June->Jan 05	
		GTFE run	>March 05	
31	Tracker flex cable	Process change implemented. Coupons from	10/15/04-	Rich
	coupon failures	flight panels failed. Steve Kahn assigned to	>11/5> 1/31	
		work with Parlex on quality and schedule.		
		Pioneer first article source inspection		
35	Reliability	FMEAs done, reviews with Subsystems	12/31/04	DiVenti
	assessments not	started. Held TKR and Mech reviews with		
	completed	SLAC, TPS, GASU and PDU held on 5/13.		
		Updates to FMEA provided on 5/21. Tony		
		distributed complete set, forwarded to		
		local subsystem managers for review		
37	SIB EEPROM	PCB approved enough parts for flight build,		Haller
	DPA Failure	still working parts for spares and qual. Have		
		enough parts for 5 boxes, all 5 will fly.		



Issues (continued)

No.	Description	Status	Due Date	Actionee
40	LAT-DAQ FPGA	SLAC to respond to AIs' from FPGA reviews		Haller
	development and	– AI's in review		
	qualification			
41	Qualification of	Still in work		Haller
	ERNI connectors			
43	TEM/TPS voltage	Combination of hardware test and modeling		Haller
	ripple	in process to determine cause and potential		
		fixes. Cause and corrective action		
		determined, retrofit in process. Replacement		
		of 2 installed TPS to be determined		



Interface Management



Interface Document Status

SC-LAT ICD ICN Status

- LAT signed this month
 - None
- Currently under signature review
 - None
- Currently in draft or revision
 - ICN-087 LAT Deliveries Table
 - ICN-0XX LAT Survey Feature Locations and Access Requirements*
 - ICN-0XX Location and Access Requirements for LAT test connectors, auxiliary cooling inlet/outlet and purge ports*
 - ICN-0XX MLI Interface
 - * Will be contained in a single ICN adding an I&T appendix to the SC-LAT ICD.

Internal LAT ICD's

- Signed off this month
 - None
- Currently in signature review
 - None
- Currently in update
 - Electronics-LAT ICD (Comments being incorporated as they are received)



Deliverables/Receivables

- LAT Deliverables
 - July: None Scheduled
 - Aug: None Scheduled
 - Sept: None Scheduled
 - Oct: None Scheduled
- LAT Receivables
 - July: SC Interface Tool, SC-LAT Flexure Pins, Bolts and Washers
 - Aug: None Scheduled
 - Sept: None Scheduled
 - Oct: None Scheduled



LAT Level Verification Status

	Verification Method				Requirements			
Category	Test	Demonstration	Analysis	Inspection	Children	Requirements		IItS
	# Complete	# Complete	# Complete	# Complete	# Complete	# Comp	Total	% Comp
Requirement Identified	-	-	-	-	-	484	484	100.0%
Flow Down Complete	-	-	-	-	-	469	484	96.9%
Draft Verification Plan	192	106	103	48	10	459	484	94.8%
Final Verification Plan	0	0	19	0	10	29	484	6.0%
Verification Plan Executed	0	0	0	0	10	10	484	2.1%
Requirement Sold	0	0	0	0	10	10	484	2.1%

- •Remainder of requirements from Spectrum Astro's 1196 EI-Y46311-000C LAT-SC Interface Control Document have been incorporated into LAT's version of DOORS.
 - •This has resulted in additional Level 2b requirements.
- •Flow down of the level 2b requirements is complete.
- •All verification methods have been incorporated into the Verification Compliance Requirements Matrix (VCRM).
- •Verification Plans are in the process of being generated.
- •Progress since last month
 - •Requirements Identified : Additional 54 Requirements
 - •Flow Down Complete: Additional 116 Requirements
 - •Draft Verification: Additional 133 Requirements
 - •Final Verification: Additional 5 Requirements
 - •Verification Plan Executed : Additional 6 Requirements
 - •Requirement Sold: Additional 6 Requirements



Key Design Metrics



Mass and Power Status Summary

- Mass
 - No change
- Power
 - No change
- FSW estimates updated
 - No change



LAT Mass Status

LAT Mass Status Report	LAT-TD-00564-11
LAT Mass Status	Effective Date: 2-Jun-05
Martin Nordby	Print Date: 29-Jun-05

Jun-05

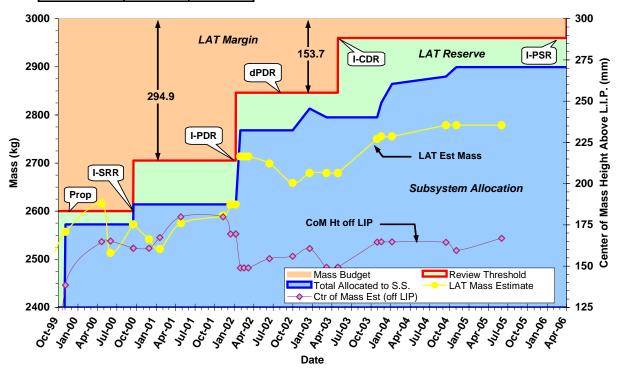
Mass (kg)	Estimate	Alloc.
TKR	523.6	530.0
CAL	1382.3	1440.0
ACD	277.6	295.0
Mech	355.7	386.6
Elec	232.0	240.0
Systems	7.5	8.0
LAT Total	2778.7	2899.6
Rsrv/Margin	221.3	
Rsrv/Margin*	8.0%	
Allocation		3000.0

^{*} AIAA G-020 recommended min reserve = 4.7% Allocations per latest mass CCB on 3 Nov 2004

Center of Mass (mm)						
CMx	-1.06	-20 < CMx < 20				
СМу	-0.87	-20 < CMx < 20				
CMz	-69.32	CMz < -51.2				
Ht off LIP	166.88	Ht < 185				

Second Moment of Inertia (kg-m²)						
lxx	1061.3	1400.0				
lyy	1013.6	1350.0				
Izz	1398.4	1580.0				

Mass Estimate Breakdown					
	%				
Parametric	56.3	2.0%			
Calculated	121.8	4.4%			
Measured	2600.5	93.6%			
Total	2778.7	100%			





LAT Power Status

LAT Power Consumption Estimate has increased by 21.7 W.

650.0

10.3%

Estimate	PARA	CALC	MEAS	SPEC
(Watts)	(Watts)	(Watts)	(Watts)	(Watts)
11.5	2.4	3.9	5.2	11.5
157.9	0.0	0.0	157.9	160.0
67.7	0.0	0.0	67.7	71.0
331.6	43.2	72.3	216.1	327.5
20.4	20.4	0.0	0.0	35.0
589.1	66.0	76.2	446.9	605.0
	11.5 157.9 67.7 331.6 20.4	11.5 2.4 157.9 0.0 67.7 0.0 331.6 43.2 20.4 20.4	11.5 2.4 3.9 157.9 0.0 0.0 67.7 0.0 0.0 331.6 43.2 72.3 20.4 20.4 0.0	11.5 2.4 3.9 5.2 157.9 0.0 0.0 157.9 67.7 0.0 0.0 67.7 331.6 43.2 72.3 216.1 20.4 20.4 0.0 0.0

- •TKR Estimate increase by 11.0 W based on average of measured modules. TKR Spec increased by 7 W.
- •CAL Estimate increase by 0.3 W based on average of measured modules.
- •T&DF Estimate increase by 10.4 W; 6.9 W due to TPS inefficiencies, 3.5 W due to flight PDU measurement.

PDR Reserve Was 15.2%
CDR Reserve Was 13.4%
Goal for PSRR Reserve > 5%

PARA - Best Estimate based on conceptual design parameters

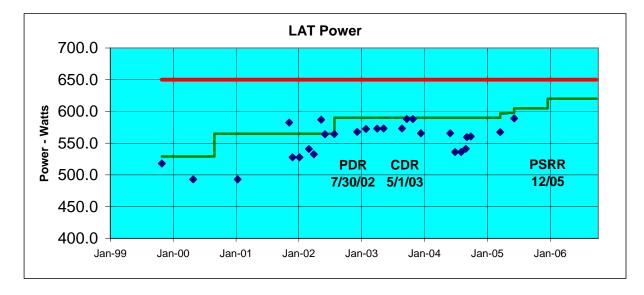
CALC - Estimate based on Calculated power from detailed design documentation

MEAS - Actual power measurements of components

Instrument Allocation

% Reserve

Goals estimated using guidelines given in ANSI/AIAA G-020-1992 "Estimating and Budgeting Weight and Power Contingencies for Space Craft Systems"





Measured LAT Power

LAT Tower Power	Unit Current (A)	Unit Power (W)
TWR 0	0.954	26.72
TWR 1	0.954	26.72
TWR 4	0.967	27.08
TWR 5	0.966	27.05
TWR 8	0.967	27.08
TWR 9	0.966	27.05
TWR Total	5.775	161.70
TWR Avg	0.963	26.95
Tower Variance	0.013	0.36

16 Tower Estimate	431.20
Orig 16 Tower Estimate	414.88
Estimated 16 Tower Increase	16.32

TEM/TPS Assembly Power Consumption					
Estimate		12.94	Watts		
Serial #	Cold	Ambient	Hot		
GLAT1754[1]					
GLAT1752	8.97	10.73	10.36		
GLAT1753	11.25	11.16	13.35		
GLAT1832					
GLAT1833					
GLAT1834					
GLAT1835		9.98			
?					
?					
?					
?					
?					
?					
?					
?					
?					
?					
Average	10.11	10.63	11.86		
LAT Estimate	161.69	170.00	189.71		
[1] Qualification Unit					

•	1. Tower measurements for bays
	0,1,4,5,8,9 are consistent and above
	predicts

•	2. Measured Tower data indicates a
	16.3 W increase in LAT power
	consumption which is supported by
	TKR, CAL, TPS unit level data
	(10.1W + 0.3W + 6.4W = 16.8W).

Estimate

Calorimeter Power Consumption							
Estimate	4.21	Watts					
Serial #	-30	25	50				
FM101 [1]	4.48	4.12	3.96				
FM102	4.59	4.22	4.06				
FM103	4.60	4.24	4.07				
FM104	4.59	4.25	4.07				
FM105	4.55	4.22	4.02				
FM106	4.61	4.21	4.05				
FM107	4.53	4.25	3.99				
FM108	4.56	4.23	4.05				
FM109	4.58	4.23	4.05				
FM110	4.61	4.26	4.08				
FM111	4.61	4.26	4.08				
FM112	4.60	4.25	4.07				
FM113	4.62	4.25	4.10				
FM114	4.59	4.22	4.06				
FM115							
FM116							
FM117							
FM118			•				
Average	4.58	4.23	4.05				
LAT Estimate	73.26	67.69	64.80				
[1]	Proto Flight	[1] Proto Flight Unit					

Serial #	Ambient
A [1]	9.70
В	9.80
1	9.87
2	9.80
3	9.90
4	9.86
5	10.15
6	
7	
8	
9	
10	
11	
12	
13	
14	
Average	9.87
LAT Estimate	157.90

Tracker Module Power Consumption

[1] Proto Flight Unit

9.18



LAT Power Status (Continued)

Survival Power

Component	Current	Subsystem Power Estimates (W)				(W)
	Alloc.	PARA	CALC	MEAS	Total	Margin
On-Orbit Average Power Total ¹	278.00	0.00	203.00	0.00	203.00	36.90%
Regulated VCHP Power Total	58.00	0.00	43.00	0.00	43.00	34.90%
Unregulated Passive Survival Power	220.00	0.00	160.00	0.00	160.00	37.50%

¹Power estimates reflect the LAT steady state orbit average. Numbers do not reflect transition into or out of survival mode, i.e. early orbit operations.



FSW Resource Usage Current Estimates

Resource	Total Available	Current Usage	Margin Factor
EPU Boot EEPROM (SUROM)	256 kB	<64 kB*	4*
SIU Boot EEPROM (SUROM)	256 kB	<64 kB*	4*
EPU EEPROM	6 MB	1.5 MB	4
SIU EEPROM	6 MB	1.5-2.5 MB	3
EPU CPU cycles	200% in 2 EPUs	40%	> 5
SIU CPU cycles	100% in 1 SIU	25%	4
EPU memory	128 MB	16-32 MB	4-8
SIU memory	128 MB	< 16 MB	8

^{*} Storing multiple copies (4 currently to use available memory) for risk mitigation



Instrument Bandwidth Resources

• LAT communication, bandwidth (BW) in Mbyte/sec

Resource	Max Total BW limited by Hardware	Max limited by SC- ground transmissi on	Ave current BW at 10 KHz max trigger rate*	Ave current BW at 2 KHz nominal trigger rate*	Margin Factor (for 10 KHz rate)
Detector to GASU-EBM	45	N/A	10	2	4.5
GASU-EBM to EPU-CPU	20	N/A	5	1	4
EPU-CPU to GASU-EBM	2.5	0.075	0.04*	0.02*	2
GASU-EBM to SIU-CPU	5	0.15	0.08*	0.015*	2
SIU-CPU to Spacecraft	5	0.15	0.08*	0.015*	2

EBM: Event-Builder Module
EPU: Event-Processing Unit
SIU: Spacecraft Interface Unit

^{*} Present performance of event filter for EPU-CPU, still being optimized. Eventually the physics filter will be adjusted/loosened to take advantage of the max average bandwidh



Key Science Performance Metrics

Parameter	SRD Value	Present Design Value
Peak Effective Area (in range 1-10 GeV)	>8000 cm ²	10,000 cm ² at 10 GeV
Energy Resolution 100 MeV on-axis	<10%	9%
Energy Resolution 10 GeV on-axis	<10%	8%
Energy Resolution 10-300 GeV on-axis	<20%	<15%
Energy Resolution 10-300 GeV off-axis (>60°)	<6%	<4.5%
PSF 68% 100 MeV on-axis	<3.5°	3.37° (front), 4.64° (total)
PSF 68% 10 GeV on-axis	<0.15°	0.086° (front), 0.115° (total)
PSF 95/68 ratio	<3	2.1 front, 2.6 back (100 MeV)
PSF 55°/normal ratio	<1.7	1.6
Field of View	>2sr	2.4 sr
Background rejection (E>100 MeV)	<10% diffuse	6% diffuse (adjustable)
Point Source Sensitivity(>100MeV)	<6x10 ⁻⁹ cm ⁻² s ⁻¹	3x10 ⁻⁹ cm ⁻² s ⁻¹
Source Location Determination	<0.5 arcmin	<0.4 arcmin (ignoring BACK info)
GRB localization	<10 arcmin	5 arcmin (ignoring BACK info)



Risk Management



Risk Management Activity

Added risk of late deliveries delaying location of system integration or performance issues.



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
Proj Mgt - 002	Moderate	If ASICs fail to meet qualification requirements; then schedule impact will occur	Focused review & test. Margin for re-runs protected where possible Individual risks Identified by subsystem Extensive use of DAQ test bed to drive out system issues	Cal/ACD ASIC's continued testing Test Bed operating No new issues
Proj Mgt - 004	Moderate	If TEM Power supply fails qualification; then final implementation may exceed schedule impacting delivery to I&T	*Key focus item identified for DAQ *TEM/PS extensive EM use as EGSE	 Implementation plan in place and proceeding Fuse audit completed Data package complete, review in process



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
SE-007	Moderate	If a critical component fails post LAT integration; then de- integration will result in cost & schedule impact	Extensive use of EM test bed to support flight H/W & S/W development Thorough qualification and acceptance tests Pre planned I&T actions for deintegration	 Qual & acceptance planning in-place I&T developing rework contingency plans. Integration plan baselined
Elec- 004	Moderate	If target hardware, requirement development or manpower is delayed; Then Flight-Software development schedule will be impacted	Detailed incremental development program Ensure sufficient software test on target hardware during development to drive out any requirement disconnects. Bring packages under CCB control Define incremental release plan to FQT	•Adapting monthly demos •Tracking EGSE resource utilization •Updated detailed test plan released •All packages in CCB 8 Aug •Completed release 4. Release 6 targeted for FQT



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
Proj Mgt - 005	Moderate	If parts and vendor orders are delayed or bids exceed expectations; then flight production costs & delivery schedule will be impacted	Manufacturing engineer added to expedite minimum cost closure Clarification and purchase package review to ensure accurate bids Increase production management staff I&T tracks parts needs	Purchase order tracking/monitoring system in place to highlight roadblocks Design documentation release plan prioritized by vendor selection and component fabrication need dates Parts needs (including long term needs) are addressed weekly during I&T 2 week lookahead meeting. MCM delivery complete
IT - 006	Moderate	If logistic or facility integration issues are found during LAT environmental test program; then re-work will delay schedule	LAT I&T to plan a roadmap of activities from LAT building 33 to completion of environmental testing LAT I&T to consider and develop opportunities to path find key activities required prior to LAT shipment to NRL	•Follow up Environmental Planning TIM held on 1 October at SLAC, I&T driving Als to conclusion •Continuing periodic TIMS, next is scheduled for 8 Sept.



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
SE - 011	Low	If individual tracker towers do not meet performance requirements due to manufacturing issues (e.g. wire bond breaks) then the LAT may not meet science requirements	Understand stability of performance to determine mitigation strategies Limit LAT temperature excursions to minimize possible propagation of some types of tracker issues Optimize placement of towers based on individual tower performance to minimize science effects and to minimize removal and replacement efforts should they become necessary	Temperature range reduced at the LAT level to allow a narrower range during Tracker and LAT tests Alternate plan for placement of Tracker A and B being implemented



ID#	Risk Rank	Risk Description	Risk Mitigation	Status
SE - 012	Moderate	If hardware deliveries are delayed (TRK, DAQ) then there will be a delay in finding system integration or performance issues	1-Improve test bed utilization 2-Early integration of ACD, 8 (TBD) towers, EM DAQ hardware, and FSW.	1-Test bed updated to accommodate calibration requirements 2-Plan in place to support early integration checkout starting in October



Cost Report

Reporting Category	Cost Incurred			Estimated Cost			Estimated Final Cost		Unfilled Orders	
	During Month Cum. to Date		Detail		Balance of	Contractor	Contract	Outstanding		
	Actual	Planned	Actual	Planned	JUL05	AUG05	Contract	Estimate	Value	
4.1.2 SYSTEM ENGINEERING										
4.1.2.1 REQ'TS MGMT, DESIGN INTEGRATION & TEST	37	29	3,144	3,170	26	30	53	3,253	3,253	0
4.1.2.3 SYSTEM ANALYSIS	0	11	1,040	1,002	10	11	-27	1,034	1,034	0
4.1.2.4 QUALIFICATION & TRACKING	109	62	647	510	56	64	-78	689	689	0
4.1.2.5 RISK & RELIABILITY ANALYSIS	0	0	99	98	0	0	-1	98	98	0
4.1.2.6 CONFIGURATION MGMT & DOCUMENT / DATA LIBRAR	5	10	296	294	9	10	7	321	321	0
4.1.2.7 MANAGEMENT & PLANNING	27	52	2,030	2,140	49	38	134	2,252	2,252	0
CAPW[3]Totals:	178	163	7,255	7,214	150	154	88	7,647	7,647	0

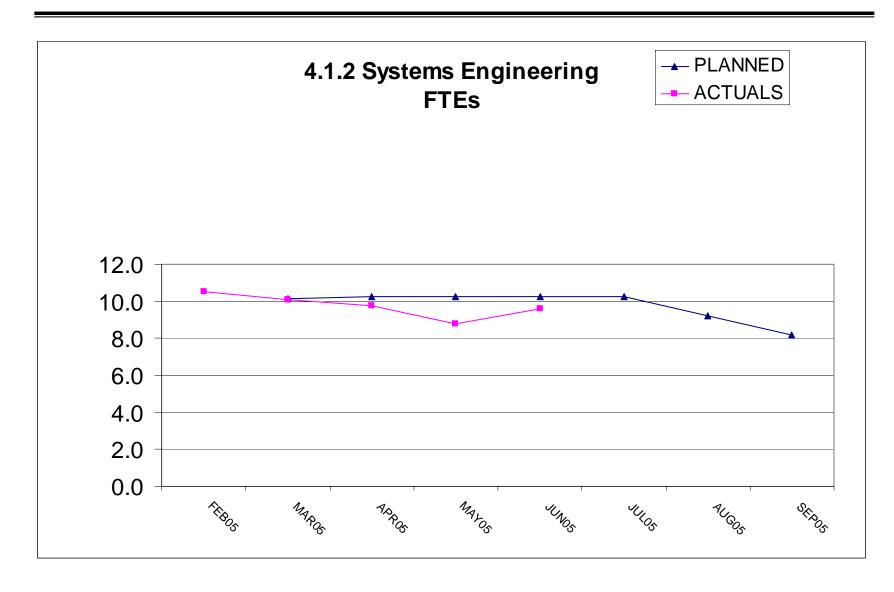


Cost Variance Explanation

- Why overrun/underrun?
 - On plan
- What will be done to correct?
 - No correction needed



FTE Report





FTE Variance Explanation

- Why overrun/underrun?
 - On plan
- What is the impact?
 - None
- What will be done to correct?
 - No correction needed